IN THE CLAIMS

Please amend the claims as follows:

Claim 1 (Original): An annular platform for the nozzle of a low-pressure turbine in a turbomachine, said nozzle having a longitudinal axis and comprising at least one fixed vane disposed downstream from at least one moving blade of a high-pressure turbine, said platform comprising a downstream portion supporting said fixed vane radially defining an aerodynamic channel which extends longitudinally between a leading edge of said fixed vane and a trailing edge of said moving blade, said platform further comprising an upstream portion extending longitudinally beyond the leading edge of said fixed vane towards the trailing edge of said moving blade so as to lengthen said aerodynamic channel.

Claim 2 (Currently Amended): <u>The A-platform according to claim 1, wherein said upstream portion includes a cooling circuit.</u>

Claim 3 (Currently Amended): <u>The A-platform according to claim 2</u>, wherein said cooling circuit includes at least one cooling cavity extending longitudinally between an upstream end of said platform and the leading edge of the fixed vane.

Claim 4 (Currently Amended): <u>The A-platform according to claim 3</u>, wherein said cooling circuit further comprises air feed means for feeding said cavity, and air exhaust means for exhausting air from said cavity.

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Claim 5 (Currently Amended): The A-platform according to claim 4, wherein said air exhaust means of said cavity comprise at least one hole opening out into said cavity and leading to the outside of said platform.

Claim 6 (Currently Amended): The A-platform according to claim 3, wherein said cavity is obtained by forming a recess in an outside portion of said platform and covering the recess with a closure device.

Claim 7 (Currently Amended): <u>The A-platform according to claim 1, constituting</u>
wherein the platform is a top platform of said high-pressure turbine nozzle.

Claim 8 (Currently Amended): The A-platform according to claim 4, constituting wherein the platform is a top platform of said high pressure low-pressure turbine nozzle, and wherein said air feed means comprise at least one orifice opening out into an air manifold for cooling the fixed vane and leading into said cavity.

Claim 9 (Currently Amended): The A-platform according to claim 8, wherein said air feed means comprise at least two orifices having different angles of inclination so as to distribute the cooling air uniformly within said cavity.

Claim 10 (Currently Amended): <u>The A-platform according to claim 1, constituting</u> wherein the platform is a bottom platform of said low-pressure turbine nozzle.

Claim 11 (Currently Amended): The A-platform according to claim 4, wherein the platform is constituting a bottom platform of said low-pressure turbine nozzle, and wherein said air feed means comprise an orifice passing through said bottom platform for exhausting cooling air from said fixed vane.

Claim 12 (Currently Amended): The A-platform according to claim 11, further comprising at least one deflector disposed parallel to said upstream portion of the platform so as to form an annular cooling channel between said deflector and said bottom platform.

Claim 13 (Original): A nozzle for a low-pressure turbine of a turbomachine, the nozzle comprising a plurality of fixed vanes supported by at least one top platform according to claim 7.

Claim 14 (Original): A nozzle for a low-pressure turbine of a turbomachine, the nozzle comprising a plurality of fixed vanes supported by at least one bottom platform according to claim 10.

Claim 15 (Currently Amended): A nozzle for a low-pressure turbine of a turbomachine, the nozzle comprising a plurality of fixed vanes supported by at least one top platform according to claim 7-1 and by at least one bottom platform according to claim 101.

Claim 16 (New): The platform according to claim 3, wherein said cooling cavity comprises baffles configured to enhance heat transfer between a cooling air and the platform.

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Claim 17 (New): The platform according to claim 1, wherein the platform is a cast piece.

Claim 18 (New): The platform according to claim 1, wherein a distance between the leading edge of said fixed vane and the trailing edge of said moving blade exceeds 80 mm.

Claim 19 (New): The platform according to claim 7, wherein the top platform has an inclination of about 30°.

Claim 20 (New): The platform according to claim 3, wherein an orifice feeding cooling air from the fixed vane to said cooling cavity in the platform is substantially aligned with a link bushing air outlet disposed at a bottom portion of the platform.

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